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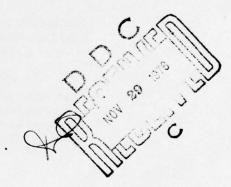
RESEARCH AND DEVELOPMENT TECHNICAL REPORT

ECOM-5804

ARTILLERY METEOROLOGICAL ANALYSIS OF PROJECT PASS

By

Abel J. Blanco Larry E. Traylor



Atmospheric Sciences Laboratory

US Army Electronics Command
White Sands Missile Range, New Mexico 88002

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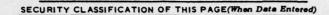
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SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered) READ INSTRUCTIONS REPORT DOCUMENTATION PAGE ECOM-5804 TITLE (and Subtitle) ARTILLERY METEOROLOGICAL ANALYSIS OF PROJECT PASS 8. CONTRACT OR GRANT NUMBER(s) Traylor Abel J. Blanco Larry E. 9. PERFORMING ORGANIZATION NAME AND ADDRESS 10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS Atmospheric Sciences Laboratory White Sands Missile Range, New Mexico 88002 DA Task No. 1L1/62111/AH71-A1 11. CONTROLLING OFFICE NAME AND ADDRESS US Army Electronics Command October 2076 NUMBER OF Fort Monmouth, New Jersey 07703 15. SECURITY CLASS. Controlling Office) UNCLASSIFIED DECLASSIFICATION SCHEDULE 16. DISTRIBUTION STATEMENT (of this Report) for public release; distribution unlimited. 17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report) 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Ballistic meteorology error Cannon ballistics Artillery meteorology Meteorological objective analysis Automated meteorological system Statistical analysis 20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The Atmospheric Sciences Laboratory, US Army Electronics Command, conducted a comprehensive experiment in ballistic meteorology during November-December 1974 at White Sands Missile Range. The purpose was to determine what improvement might be made in the representativeness of meteorological messages furnished to artillery batteries by collecting all meteorological data available in a corps size area, performing a relatively simple analysis, and disseminating the results to the batteries. One thousand rounds of 8-inch howitzer ammunition were

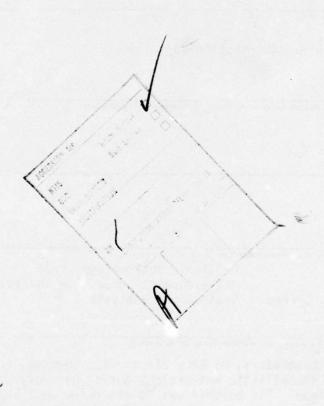
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20. Abstract (cont)

fired over a 5-week period with concurrent meteorological data taken at eight sites on the Missile Range. The most striking feature of the results is the relatively small deviation in predicted impact (less than 40 m range sigma) due to meteorological error in 85% of the 79 firing series analyzed.



CONTENTS

	Page
INTRODUCTION	2
APPROACH	2
EXPERIMENT DESCRIPTION	3
EXPERIMENTAL ERROR	7
ANALYSIS	9
METHOD I	10
METHOD II	11
METHOD III	12
RESULTS	13
CONCLUSIONS	19
REFERENCES	23
APPENDIX ALL HARDWARE PARAMETERS AND ACTUAL AND SIMULATED DISPLACEMENTS	24

INTRODUCTION

The accurate assessment and application of atmospheric wind, temperature, and pressure are well-known essentials to effective delivery of artillery munitions. The belief that meteorological errors are major contributors to the total delivery error budget is widely held and is supported by many previous studies of the problem [1]. These errors may be broadly categorized as instrumental and environmental. Environmental meteorological errors in this context are defined as those arising from the separation in time and space between measurement and application.

The Atmospheric Sciences Laboratory (ASL) is conducting studies designed to minimize the above meteorological errors. As part of these studies, a field experiment (PASS - Prototype Artillery Subsystem of the Automatic Meteorological System) involving actual howitzer firings together with meteorological data collection at several sites on White Sands Missile Range (WSMR) was conducted during November-December 1974. This report addresses the analysis of the ballistic meteorology data from that experiment, the results thereof, and their implications. A brief description of the experimental setup is included for completeness. A comprehensive description is available under separate cover [2].

APPROACH

The main purpose of the ballistic meteorology portion of the PASS experiment was to test the hypothesis that an analysis of the upper air soundings available from five to six meteorological sections (the assumed usual number in a type corps) could substantially reduce the contribution of meteorological errors to the total delivery error budget. Relatively simple objective analysis algorithms were chosen as the vehicle for combining the multiple soundings since the computation time and computer core storage available for the purpose are limited for field application.

To compare the merits of several competing algorithms via a firing experiment, two general procedures for the experiment are feasible. One would be to identify a number of "most promising" algorithms in advance and conduct a series of firings at the same target by using each algorithm in turn to produce a meteorological message for the firing problem solution. The algorithm giving the smallest dispersion about the target center would be considered the "best" of the group. To gather a sufficiently large sample for statistically significant results in this manner would require a prohibitive amount of ammunition and time for even a small number of candidate algorithms. The second approach would be to conduct a series of firings at the same aiming point by using any reasonably accurate meteorological message in the firing problem solution. Thus, the range to impact should not vary enough from series to series to cause significant changes in unit wind effects, unit muzzle velocity effect $\frac{(\partial R)}{\partial MV}$, and so on. Then by redefining the mean point of impact of each series as the target center for that series, computer trajectory

simulations could be examined to determine the relative ability of any number of candidate algorithms to predict the observed mean point of impact (Figure 1). The algorithm yielding the smallest dispersion about the observed mean impact points would then be the "best" of those examined. This second approach was chosen for the PASS experiment since (1) it was considerably more economical in materials and time, and (2) it was not certain that the most promising objective analysis algorithms had been identified.

To further clarify the approach taken, suppose that a given series is fired and the mean point of impact is 75 m over the nominal range to the aiming stake and 35 m to the right (Figure 1). Suppose also that the muzzle velocity, firing angles, and other nonmeteorological variables of this series were accurately measured. Then any meteorological message (together with the measured hardware variables of this series) which produces a simulated mean impact point identical to the actual impact point may be regarded as the most desirable meteorological message for that series.

It is recognized that the transformation (via the ballistic equations of motion) of zonewise variables of pressure, temperature, and wind components into an impact point represents a transformation from E_n to E_2 (n > 2) and is not unique in this case. However, if a meteorological algorithm leads to simulated impacts consistently closer to actual impact (i.e., smallest dispersion) than the other candidate algorithms, and if this is a statistically significant result, then it may be concluded that the best algorithm of the group has been isolated.

EXPERIMENT DESCRIPTION

Two 8-inch howitzers were emplaced on the Missile Range, and a well-qualified crew from Fort Sill fired them into an impact area which was cleared and leveled for several hundred meters around the aiming stake. The firings were comprised of 8-round series (10 rounds for the first series of the day), 2 minutes between each round, 1 hour between each series. The experiment spanned 20 days (not necessarily consecutive). Guns were alternated daily whenever possible, with the number of series fired each day varying from 4 to 10.

Impact locations for each round were determined by triangulation from three flash ranging stations placed symmetrically around the aiming stake [3]. Muzzle velocity of each round was obtained from a Doppler velocimeter supplied and operated by WSMR.

Upper air soundings were taken at ten sites during the experiment, but only eight of these were intended for use in the ballistic analysis (see map [Figure 2] and Table 1 for relative locations). Five stations simulating the meteorological sections in a type corps released balloons simultaneously each 2 hours, beginning 1 hour and 15 minutes before the first series of the day. The remaining three stations released simultaneously 1/2 hour after the first five. The above measurements were

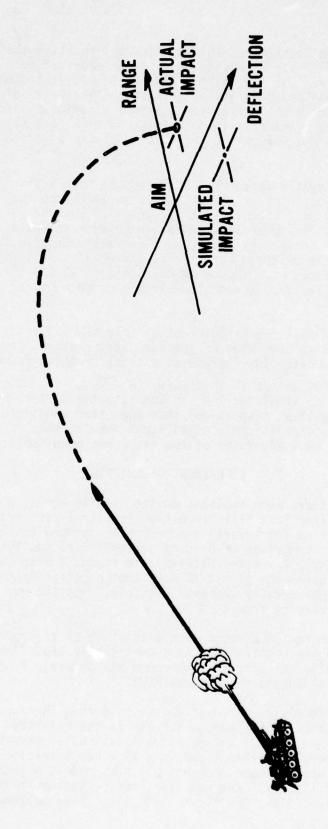
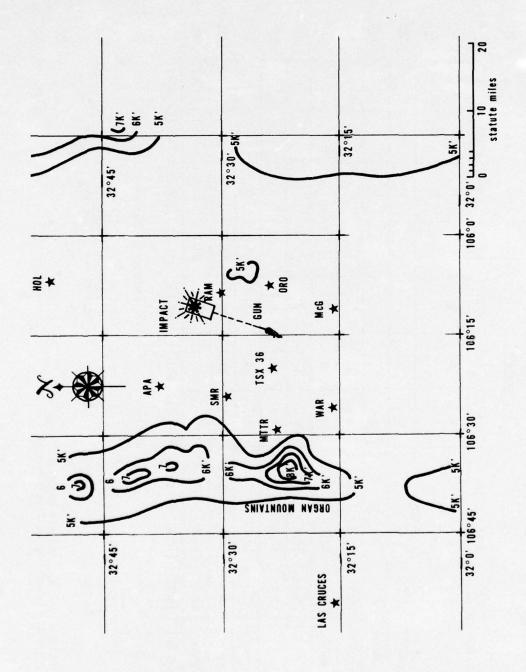


Figure 1. Pass statistical analysis.



Approximate map of artillery meteorological sections, gun position and impact area for artillery meteorological comparisons by USAECOM Atmospheric Sciences Laboratory at White Sands Missile Range, Nov 74. Figure 2.

TABLE 1. DISTANCE BETWEEN RAWINSONDES OR SIMULATED ARTILLERY METRO

DISTANCE BETWEEN RAWINSONDES

(kilometers)

		TSX	ORO	TSC	MCG	WAR	MTR	SMR	RAM	APA	HMS	
HMS	10										0	(51.7)
APA	6									0	37.8	(26.9)
RAM	œ								0	25.0	39.4	(13.7)
SMR*	7							0	24.4	16.2	51.7	(17.4)
MTR	9						0	10.9	33.2	26.2	62.6	(22.6)
WAR*	5					0	15.9	21.9	34.0	37.9	70.2	(20.3)
MCG*	4				0	20.8	32.3	31.5	25.7	43.0	65.0	(16.2)
TSC	3			0	68.0	47.2	41.7	51.5	75.0	62.6	100.3	(63.5)
0R0*	2		0	73.6	15.6	28.7	33.0	27.0	10.7	33.0	49.6	(10.4)
TSX*	-	0	16.5	57.5	19.5	16.5	16.4	12.1	18.3	24.5	53.8	(6.2)
LON		1. LC-36*	Orogrande*	Las Cruces [†]	McGregor*	War Road*	MTTR+	SMR*	Rampart	Apache	Holloman	(Gun Site)
SECTION		-:	2.	 	4.	5.	.9	7.	8.	9.	10.	

13.1 mi 7.5 mi 19.6 mi 37.3 km 10.7 km 100.3 km n n n 23.2 mi 6.6 mi 62.3 mi Mean Distance Minimum Maximum

All Stations

Simulated Corps Stations 21.1 km 12.0 km 31.5 km 11 11 11

*Indicates those used in simulated corps. †Indicates those omitted from ballistic analysis.

obtained by using a rawinsonde with standard AN/GMD-1B tracker, and computer meteorological messages were prepared in standard artillery fashion [4]. Wind-only profiles were obtained each hour at all eight stations by modified T-9 radar equipment, coincident with the rawinsonde measurements every second hour. The winds were incorporated into separate computer meteorological messages utilizing either concurrent or 1-hour-old GMD determined pressures and temperatures as appropriate. The howitzers, meteorological stations, aiming stake, flash ranging positions, and velocimeter antenna were all located by survey crews furnished by WSMR. Coordinates were given in Universal Transverse Mercator (UTM), White Sands Transverse Mercator (WSTM), and White Sands Cartesian System (WSCS). WSTM was selected for use in the analysis.

The procedure of laying the howitzer for any given series of rounds was predicated solely on the range safety requirement of impacts well within the cleared zone designated as the impact area, so that the coordinates of the center of this area represented the aiming point or "target" so far as the Fire Direction Center (FDC) was concerned. For more details, see [2].

EXPERIMENTAL ERROR

The major variables of importance to the ballistic meteorology analysis are, series by series:

- 1. The meteorological data
- 2. The mean fall-of-shot location
- 3. The quadrant elevation and azimuth angles
- 4. The mean muzzle velocity
- 5. Projectile weight
- 6. The exterior ballistics parameters of the projectile (drag coefficient, etc., assumed to be identical for identical projectiles)

Of these, the last is assumed to be subject to bias error only, while both bias and random errors may occur in the other five.

Several tedious hand editing passes through the "raw" meteorological data by separate groups within ASL have revealed numerous errors of presumably human origin in the temperature and pressure values, with subsequent influence on the wind determination at those points; but this has not proven to be a significant problem to the ballistic analysis. The smoothing inherent in averaging over height layers several hundred meters thick plus the smoothing inherent in the ballistic equations of motion tends to minimize random errors of both instrumental and human origin. In fact, the meteorological data used in the ballistic analysis is

identical to that collected in the field with the exception of a visual inspection of the computer meteorological messages for obvious spike errors in the temperature and pressure profiles. Errors of this type could and should be easily detectable by competent military personnel in the field.

The location of the impact point of each round was examined, and those rounds wherein the error was considered excessive were discarded. This procedure occasionally caused an entire series to be discarded. The total losses were few and were caused by commencing the day's firings just before dawn when the flash rangers had difficulty seeing the burst point. A detailed treatment of this portion of the experiment is given in [3], but essentially the impact locations retained for analysis should be accurate to $\pm 10\,$ m.

Firing angles were set into the howitzers by the gun crew referencing a surveyed orienting line for azimuth and checking the quadrant elevation setting with a gunner's quadrant. Since the howitzer had to be depressed to load, the elevation angle was set and checked for each round, but the azimuth laid was checked only after completion of the series. Twenty projectiles, selected at random and weighed, proved to be well within tolerance of their four-square weight. Twenty powder bags (charge 7 white) were also weighed with similar results. The projectiles and charges were from the same lot, respectively. The firing crew were extremely careful in loading, using the hydraulic ram for all but a few rounds, and employing a gauge to insure that the charge was always placed in the same position in the breech. Further, the charge stack was sheltered on three sides and the top from the sun and wind, with propellant temperatures recorded at one top and one bottom corner of the stack. These two temperatures were averaged to obtain the propellant temperature used to solve the firing problem.

The measurement of muzzle velocity was the most difficult of the required variables. The velocimeter was emplaced approximately 100 m behind the guns, with a less than optimum look angle. The method of reducing the data to muzzle velocity used in the first few days of the experiment was the same as that used to determine rocket velocity, which reaches its maximum at a relatively large distance from the launcher. The results were therefore more indicative of radial velocity component than total velocity and were too low. When notified of this error, the velocimeter branch derived and applied corrective factors (to account for the unsatisfactory geometry) which subsequent analysis has shown to be of completely acceptable accuracy and the best of several other means investigated for transforming radial velocity to true velocity. The precision of the velocimeter data is excellent, as evidenced by extremely high correlation between the round-to-round difference in measured muzzle velocity and the round-to-round difference in range components of impacts, both before and after the corrective action to reduce the bias.

ANALYSIS

The objective of the data analysis was to isolate the effect of errors in ballistic meteorology on the impact points of the projectiles and to examine some methods of reducing the error. The procedure is best illustrated by the following set of linearized ballistic equations. Only the range component of impact location will be shown, but analogous equations for the cross component are easily written. Let the true mean range to impact (R) for a given series be represented by:

$$R = R_N + \Delta V + M + NEGLIGIBLE TERMS$$
 (1)

where R_N is the nominal range which would be reached at the true quadrant elevation angle if muzzle velocity and meteorology effects are standard. ΔV and M are the true range displacements from R_N (in meters) due to nonstandard muzzle velocity and nonstandard meteorological conditions. The negligible terms include nonlinear or second order effects of smaller magnitude.

The relevant quantities in Eq. (1) measured (or calculated) during the experiment are subject to error and may be written as:

$$R_{M} = R + \varepsilon_{R}$$

$$R_{N\Theta} = R_{N} + \varepsilon_{\Theta}$$

$$\Delta V_{M} = \Delta V + \varepsilon_{V}$$

$$M_{A} = M + \varepsilon_{A}$$
(2)

The various errors are ϵ_R , due to imprecise location of fall-of-shot; ϵ_θ , due to quadrant elevation error and written as an effector of nominal range; ϵ_V , due to measurement error in muzzle velocity; and ϵ_A , due to departure from the "true" meteorology of a meteorological message derived with algorithm A, containing instrumental error, algorithm error, and natural space-time variability. The errors are assumed to be random from series to series, to be drawn from populations of zero mean with variances characteristic of the PASS experiment, and to be statistically independent.

Ballistic trajectory simulations were made series by series by utilizing the series mean measured muzzle velocity, quadrant elevation angle, azimuth angle, standard (four squares) projectile weight, and various meteorological messages produced by the competing meteorological analysis

algorithms. Expressing the simulated impact range in the simple linear fashion of Eq. (1):

$$R_{A} = R_{N\theta} + \Delta V_{M} + M_{A}$$
 (3)

Subtracting Eq. (3) from R_M

$$\Delta_{A} = R_{M} - R_{A} \tag{4}$$

From Eqs. (2) and (1):

$$\Delta_{A} = -\epsilon_{\theta} - \epsilon_{V} + \epsilon_{R} - \epsilon_{A}$$
 (5)

Lumping the nonmeteorological terms together and labeling the sum "experimental error,"

$$\Delta_{A} = \varepsilon_{EXP} - \varepsilon_{A} \tag{6}$$

The variance of $\Delta_A(\sigma^2_{\Delta A})$ is a measure of the performance of meteorological algorithm "A" in reproducing the observed impacts. This is subject to the assumption that the estimators of meteorological displacement are unbiased estimators ($\bar{\Delta}_A$ = 0); otherwise, a better measure might be the RMS Δ_A . Further, it is meaningful to compare $\sigma^2_{\Delta A}$ and $\sigma^2_{\Delta B}$ for algorithms "A" and "B" to discover if any statistically significant difference exists and, if so, which is better (smaller).

The meteorological algorithms to be tested on the PASS data were limited to three, hereafter designated as Methods I, II, and III. Method I is the single station technique currently in use by the Field Artillery [4]. Methods II and III were selected from a group of candidate objective analysis schemes based on comparisons between estimates of ballistic meteorology messages given by the schemes and by an actual sounding at the place and time of estimation. The candidate group was by no means an exhaustive collection of available objective analysis techniques. The initial screening criteria to form the group were based principally on simplicity. The two methods selected from the group will be defined and briefly discussed in this report, but a more complete discussion including the other candidates is given in [2].

METHOD I

The station designated "TSX" (Figure 2) was selected as the Method I station. It was the closest to the howitzers and therefore the most likely to be chosen by a commander in the field to provide computer

meteorological messages for his battery. The soundings were taken according to standard artillery meteorological methodology [4] through line 9 (4000 m AGL). This practice was followed at all stations.

METHOD II

This algorithm takes the form (suggested by Barnett, ASL) of a linear predictor, or "weighted average,"

$$\hat{A} = \Sigma \alpha_i A_i$$

where

$$\Sigma \alpha_i = 1$$

and

$$\alpha_{i} = \alpha 1 / (c_{1} d_{i}^{\frac{1}{2}} + c_{2} d_{i}^{\frac{1}{2}})$$

(all summations taken over the five "corps" stations). A is a measured meteorological message parameter of interest (wind component, pressure, or temperature) at the ith station; the weights α_{i} are inversely proportional to a function of the distance (d and time (t separation between measurement and application (station location and time of release); and A is the estimate of the parameter. The weighting was performed zonewise, with no dependence on zones above or below.

The intuitive appeal of this particular estimator is better illustrated by considering that the weights should be proportional to the confidence that the measured parameter represents the actual parameter at the time and place of application. Other investigators [5] have described atmospheric time-space variability in the following manner:

$$\sigma_d = c_1 d^{\frac{1}{2}}$$
 (variability over distance d),

$$\sigma_t = c_2 t^{\frac{1}{2}}$$
 (variability over time interval t).

These relationships then give rise to the Method II estimator form. The value of c_1 was obtained from space variability data furnished by personnel of the US Army Ballistics Research Laboratory (BRL) using a least squares fit [6]. The value for c_2 was then obtained from $c_1 = c_2/\sqrt{30}$, i.e., a distance of 30 km gives equal variability to a time separation of 1 hour. This represents a compromise between reported variability

equivalences of as much as 46 km/hr to as little as 12 km/hr. The specific values are:

 $c_1 = 0.47$

 $c_2 = 0.3189$,

distance in kilometers, time in minutes.

METHOD III

Although Method II has a time dependent term, it is in no way intended to extrapolate time (or space) trends to the time and place of firing. Method III is an attempt to detect both time and space trends and extrapolate to the applicable location and time.

At each station, the meteorological message was extrapolated to the time of firing according to the following rules:

- 1. If only one message was present in the data bank for a station, persistence was invoked (no change).
- 2. If two messages were available for a station, a linear trend was extrapolated to the firing time, the average of the two messages was calculated, and finally the mean of the linear trend value and the average value was obtained.
- 3. If more than two messages were available from a station, the procedure was identical to step 2 except that a cubic spline was fit to the data points instead of the linear trend.

The procedure was done zonewise for each station; and in all cases analyzed, the requirement was invoked that the most recent release at a station would be no more than 135 minutes before firing time or the station would be ignored.

Having obtained a message at each station extrapolated forward to a common time, the next step was to fit a least squares plane to the data in space and evaluate the plane at the howitzer location, thus

$$\hat{A} = ax + by + c$$
.

A is the estimate of a meteorological parameter; x and y are the howitzer coordinates; and a, b, and c are determined from the least squares fit over the stations. A plane was fit for each atmospheric zone and for each parameter of interest (wind components, pressure, and temperature).

RESULTS

After editing for fall-of-shot location errors, velocimeter dropouts, etc., the original 115 firing series were reduced to 79 which were suitable for analysis. Table 2 summarizes the analysis of the 79 cases. The data were partitioned into a set of 68 cases labeled "normal" and a set of 11 cases labeled "special" when it was noticed that relatively few of the cases gave large errors. The selection of 100 m range miss-distance as the partition was not entirely arbitrary, since the miss-distance frequency histogram indicates a bimodal distribution with the point of overlap of the modes being approximately 100 m. Figure 3 represents the same data as Table 2 but in graphical form.

There are two major features of the results in relation to the objectives of this experiment. First, an attractive decrease in ballistic meteorology error failed to materialize for the algorithms tested. When it is recalled that experimental error is of necessity included in the results, it is apparent that no algorithm will offer much improvement in the "normal" set, since the meteorological error is already small. An examination of all the experimental data inputs leads to the conclusion that time trends in the meteorological regime produced the "special" set of 11 cases of large miss-distance. Figures 4 and 5 illustrate time-trending winds on 2 of the days where 100 m range misses were indicated by the simulation. The meteorological message zone 5 (1500-2000 m AGL) vector winds reported by all operational stations are depicted (vector origin at station) and at times exhibit distinct changes in speed and/or direction occurring over time periods of the order of 2-3 hours. These changes are in the proper direction and of the approximate magnitude to produce the observed miss-distance in range, while leaving the deflection miss-distance relatively small. The orientation of the guntarget line to the prevailing winds (westerly) during the test coupled with the general tendency of the time trends to proceed from crosswind to head wind produced the greatest error in the range component of impact. This is most likely an accidental circumstance peculiar to the PASS experiment. It remains possible then to effect significant improvement in meteorological error for these cases, although Method III failed to do so.

Second, the low percentage (15%) of the very difficult "special" cases is itself interesting. Similar experiments made by other investigators at different locations and seasons [7,8] corroborate this percentage, indicating that the meteorological conditions encountered during the PASS experiment were neither abnormally quiet nor noisy. The frequency of occurrence of time trending meteorological situations is obviously a strong function of season at a given place, but the quirks of local climatology preclude any general statement about the functional dependence on latitude or other purely geographical variables.

The space variability depicted in Figure 6 was computed by using Method I for each station in turn. The stations known as McGregor (MCG) and WAR (Figure 2) were situated to the east of the southern end of the Organ Mountains and are believed to have been adversely affected by the disturbance of the prevailing westerly flow around the mountain. The time

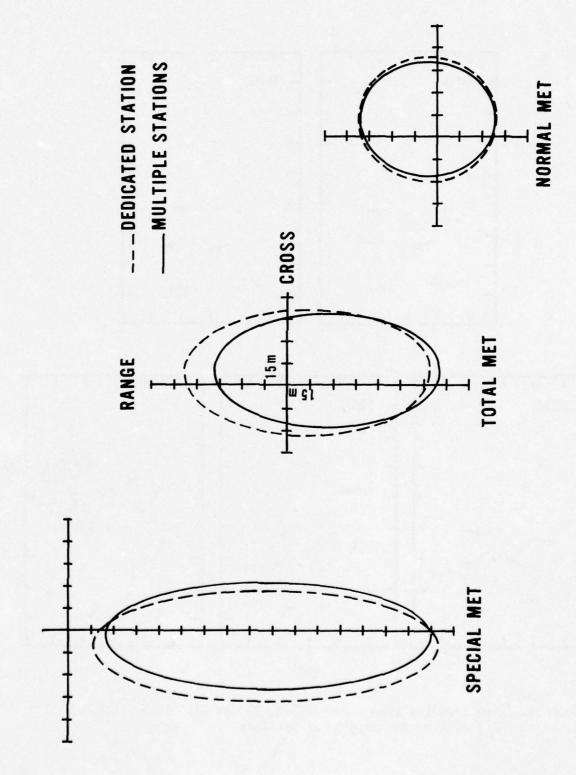
TABLE 2. COMPARISON OF RESULTS FOR METHODS I, II and III

Total of 79 Cases Analyzed (1-2 hr GMD)

$\frac{Rang}{\Delta R}$	e (m)		Deflect	ion (m)
ΔR	σΔR	Meteorological Message	ΔD	σΔĎ
-13	69	Method I	9	36
-27	63	Method II	10	33
-21	63	Method III	12	33

Partitioned Statistics from PASS (1-2 hr GMD)

		Actual - Si	mulated			
Method	Partition	No. of Series	$\frac{Range}{\Delta R}$	e (m) σΔR	$\frac{Deflection}{\Delta D}$	(m) σΔD
I	Special	11	-130	98	-11	31
	Total	79	-13	69	9	36
	Normal	68	6	38	12	36
II	Special	11	-132	91	-5	30
	Total	79	-27	63	10	33
	Normal	68	-9	35	12	33
III	Special	11	-127	93	0	29
	Total	79	-21	63	12	33
	Normal	68	-4	35	13	44



One probable error difference eclipse (actual - simulated). Figure 3.

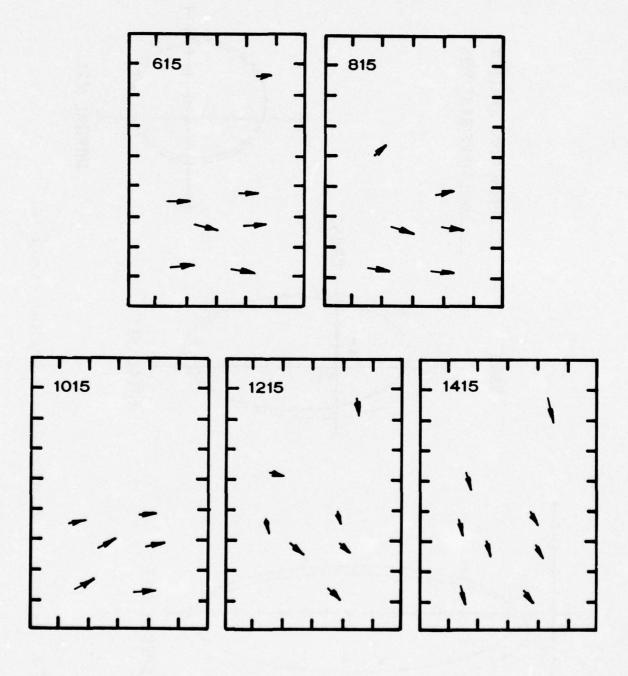


Figure 4. Time trending wind vector field, 23 Nov 74. Zone 5 (1500-2000 m AGL) with vector origins at stations.

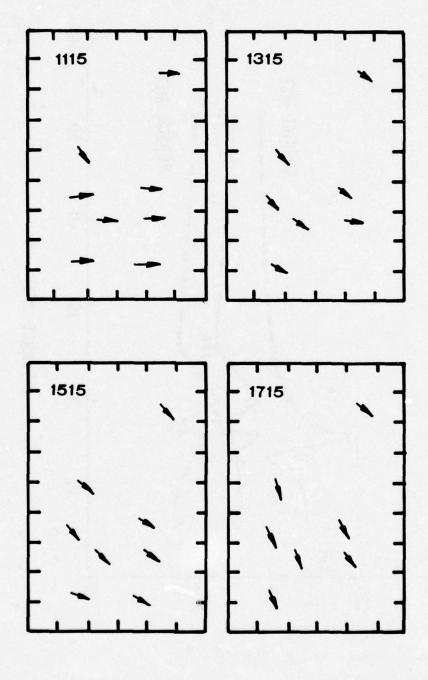


Figure 5. Time trending wind vector field, 5 Dec 74. Zone 5 (1500-2000 m AGL) with vector origins at stations.

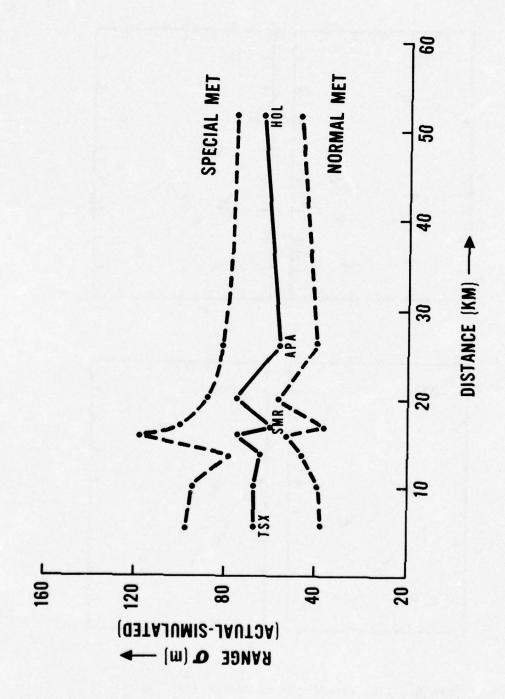


Figure 6. Space variability for 1-2 hr met.

variability curve of Figure 7 is self-explanatory. A comparison of Figure 7 with Figure 6 indicates the dominance of time variability over space variability insofar as ballistic effects are concerned. The SMR station is shown because it produced the best results of any single station (Method I) and was also located essentially upwind from the trajectory for most of the firings. Table 3 presents the combinations of soundings and firings utilized to obtain Figures 7 and 8.

CONCLUSIONS

The analysis of ballistic data from the PASS experiment indicates that the experimental errors were well within acceptable levels from the standpoint of ballistics. The application of two unsophisticated meteorology analysis algorithms to the data obtained from five meteorological stations failed to produce a practically significant decrease in the meteorological contribution to the total delivery error budget. In 68 of the 79 firing series examined, the error due to meteorology was less than 100 m range miss; and in fact, the dispersion for these 68 cases was small (~ 37 m range and ~ 35 m deflection), which includes experimental error. This point should be examined closely in any future attempt to reduce meteorological disperions, since further investigation may reveal that the percentage of meteorological conditions wherein significant improvement is possible is too small to be of importance.

The idea that time variability of meteorology (in particular wind) is the major factor in large ballistic meteorology errors was corroborated and reinforced by the PASS results. The mode of simultaneous atmospheric soundings did not allow a study to be made of the value of soundings staggered in time from station to station over an interval of 2 hours, but the implication is clear that such a release schedule would go far toward reducing meteorological errors due to staleness. A central collection and disbursing system would therefore be useful in disseminating the most recent sounding to all batteries. In addition, such a system could make simple discrimination decisions such as separating stations from batteries where large terrain features might intervene, etc.

A final conclusion is that most of the advantage to be gained from a central disbursing system for artillery meteorological data would stem from the steady flow of fresh (½ hour to 1 hour old) meteorological messages to any battery, uninterrupted by relocation of a meteorological section, mechanical breakdown of the section, enemy action against the section, or any of the myriad hazards surrounding the prompt delivery of information from a given artillery meteorological section. ASL is currently making a detailed study of what may be achieved in increased artillery effectiveness from this viewpoint.

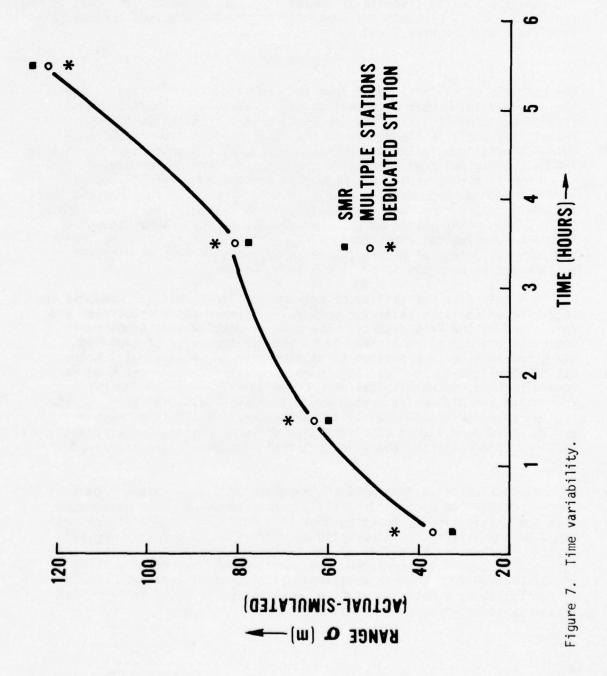


TABLE 3. SIMULATED IMPACTS VERSUS CORRESPONDING ACTUAL IMPACTS FOR VARIOUS METEOROLOGICAL AGES

Date	0.3	1 1	orolo 2	gica 3	Meteorological Age (hr 1 2 3 4	Pr.	19	Date	0.3		orolo 2	gical 3	Meteorological Age (hr)	hr 5	9	Date		0.3	Meteorological	2 2	al Age	ا	2
Nov 8	1500	1400*	× ×	*	×			Nov 18	0630	0730	× ×	×	*			Nov	Nov 27	1130	1030	× ×	×	×	
II vov	0700	0800	*	*				91 vov 19	0630		× 0630+ × 0830	*	*			Dec 2		0730 0930 1130 1330	0630 0830 1030 1230	* * * *	× × ×	× × ×	
Nov 12	0630	0530	×	×				Nov 20	1400	1300	× ×	×	×			Dec 3		1000	0000	× ×	×	×	
Nov 14	0600 0800 1000 1200	0500 0700 0900 1100	× × × ×	× × ×	× × ×	* *	× ×	Nov 23	0830 1030 1230+ 1430+	0730 0930 1130+ 1330	× × × ×	× × ×	* * *	× ×	× ×	Dec 5		1330+	1230+ 1430+ 1630+	× × ×	× ×	* *	
Nov 15	0615 0815 1015 1215	0715 0915 1115	× × × ×	× × ×	× × ×	× ×	× ×	Nov 26	1430	1330	× ×	×	*			Dec 7		0730 0930 1130 1330	0630 0830 1030+ 1230 1430	× × × × ×	* * * *	* * *	

X - Simulated impacts
* - Time of actual firing
+ - Firings with > 100 m range miss

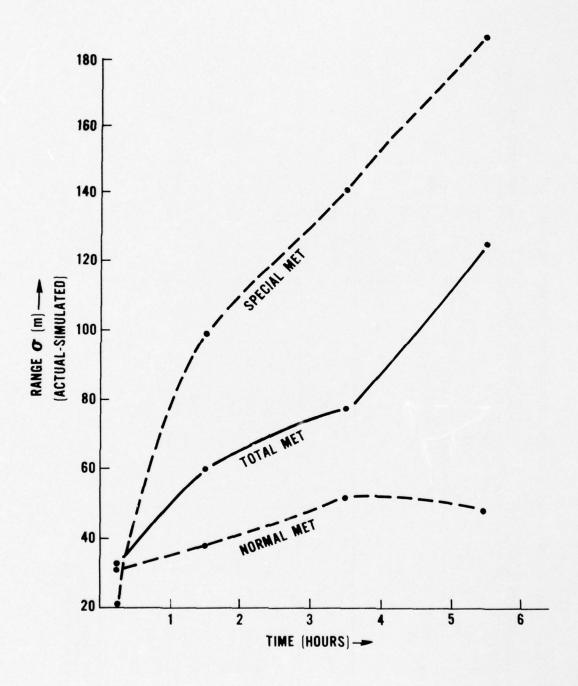


Figure 8. Time variability for SMR (generally upwind station).

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OMNITAB PASS DATA

APPENDIX. ALL HARDWARE PARAMETERS AND ACTUAL AND SIMULATED DISPLACEMENTS

SERIES	1.00000	2.00000	3.00000	4.00000	5.00000	6.00000	7.00000	8.00000	9.00000	10.0000	11.0030	12.0000	13.0000	14.0000	15.0000	16.0000	17.0000	18.0000	19.0000	20.0030	21,0000	22.0000	23.0000	24.0000	25.0000	20.0000	27.0000	28.0000	29.0000	30.0000	31.0000	32,0000	33,0000	34.3000	35,0000	36.4000	37.0000	38.0000	39.0000	40.000	41.0000	42.0000	43.0000	44.0000	45.0000	46.4000	47.0000	48.0000	49.0000	50.0000
	65,0000	64.0000	62,0000	62,0000	50.0000	51,0000	48.0000	62.0000	29,0000	50.0000	63,0000	58,0000	60,000	60.0000	53,4000	54,0000	55,0000	52,0000	53.0000	48.0000	51,0000	52,0000	51,0000	51,0000	49.0000	58,0000	57,0000	5.000	62,0000	61.0000	61,0000	70,0000	68,0000	75,0000	70,0000	62,0000	59.0000	62.0000	0000.09	62.0000	62,0030	60.0000	61,0000	0000.99	64.0000	0000.09	0000.99	58,0000	59.0000	0000.09
> 1 5	596.830	596.950	596.810	597.250	593.620	593.750	592.360	593.300	295.540	286.220	595.650	595.150	000.000	294.440	593,300	592.570	390.830	593.370	340.040	290.000	600.170	592.430	597.100	593.340	592.430	393.370	240.730	244.720	294.160	070.070	592.160	597.430	597.400	598.340	590.360	598.770	247.540	295.900	045.646	592.900	594.820	594.070	593.270	598.740	590.840	298.400	597.030	596.300	297.960	296.830
9.6	u11.000	412,000	407.000	409.000	443,000	452,000	440.000	440.000	439,000	000.04	428,000	432,000	428.000	25	432,000	430.000	434,000	437,000	43: 000	141.000	424,000	408.000	414,000	416.000	417.000	407.000	000 - 10	407.000	417,000	443.000	422,000	411,000	415,000	411,000	414.000	414.000	000.414	419.000	441.000	418.000	433,000	439,000	437.000	406.000	431.000	426.000	426.000	423.000	414.000	411.000
7.4	398.000	394,000	394.000	407,000	394,000	394.000	394.000	398.000	399.000	000.045	394,000	395,000	395,000	395,000	395,000	396.000	396.000	395,000	2000.545	243,000	393,000	393,000	392,000	391,000	391,000	396,000	346.000	396.000	386,000	000.180	388.000	402.000	401,000	401.000	401.000	8	387,000	367.000	0	387.000	391,000	392,000	393.000	393,000	392.000	393.000	393.000	401.000	402.000	401.000
1146	1400.00	1500.00	1611.00	1706.00	607.000	200.000	800.000	534.000	430.000	000.00	500.000	000.009	700.000	800.000	900.000	1000.00	1100.00	1200.00	1000	000.417	815.000	915.000	1015.00	1124.00	1215.00	610.000	930.000	430.000	090.054	000.000	930.000	1302.00	1400.00	1500.00	1600.00	230.167	930.000	730.000	90.7501	1130.00	1230.00	1330.00	1432.00	1332.00	1434.00	1524.00	1430.00	1030.00	1134.00	1630.00
M35G			11083.2	11084.2	111111	11112.1	11113.1	211711	11122.2	716316	_	11142.1	11143.1		11145.1		1.147.1	11152.1	11153		=	11155.1	11156.1	11157.1	11158.1	11182.2	: :	: :			11194.1	11201.2	11202.2	11203.2	11204.6	11,11	11333	11234		1.555.11	11236.1	11237.1	11234.1	11261.2	11262.2	21593.5	11264.2	110711	11272.2	3161311

SERIES		51.0000	52.0000	53.0000	0000-45	55.0000	0000	00000	57.0000	58.0000	59.0000	0000000		61.0000	62.0000	63.0000	0000	0000	00000.50	00000.99	67.0000	68.0000	66.000	70.000		71.0000	72.0000	73-0000	0000	000001	75.0000	76.0000	77.0000	78.0000	19.0000
4		010000	56,0000	61,0000	61,0000	54.0000	56.0000	0000	000000	23.0000	52,0000	0000.09		24.3030	52.0000	48,0000	63.00.00	0.00	80.000	0000000	62,0000	62,0000	62.0000	46.0000		46.0000	43,0000	42.0000	42010	2000	9000.	48.0000	50.0000	49.0000	52,0000
> × + F	504 40A	274.120	002.120	2940.040	600.700	600.360	597.470	602.310	0004		246.350	061.709	An2 020	0000	005.466	297.510	601.780	602.430	25.	077.000	003.230	247.720	599.630	597,333		204.000	596.330	600.540	593.200	600.730	05.00	0100110	018.765	594.740	591.860
36	403.000	423-000	000.621	000.01	200.000	000.414	417.000	419,000	412.000	0000-11-1	2000	000.221	416.000	000	000.	117.000	407.000	414.000	421.000	413-000	000	000	455.000	448.000	431 000	2000	150.000	431.000	425.000	429,000	425-000	25.000	123,030	146.000	429.000
24	401,000	396.000	304 000	200	200.100	000.	347.000	396,000	392,000	396.000	307 000		397,000	397.000		000.	341.000	390.000	384.000	384.000	307 000		000.185	345.000	201 000	000	000	200.	344,000	394,000	393.000	391.000	000	000.	244,000
TIME	1330.00	432.000	732.000	A30.000	030-060	1034.00	00.	1130.00	1232.00	1330.00	700.000		800.000	900.000	1000-001	1330.00	10.0031	1330.00	1435.00	1530.00	1630.00	1730.00	130 000	000 · nc •	730.000	430.000	00000	1032.00	10.250	1130.00	1230.00	1334.00	1430.00	25.28.00	00.
980#	11274.2	12021.2	12022.2	12023.2	12024.2	12025.2	13034 3	7.63031	1.0027.2	12028.2	12031.2		12032.2	12033.2	12034.2	12051.2	12062 2	7036031	12053.2	12054.2	12055.2	12056.2	12071.1		12072.1	12073.1	12074.2	12075.2		7.6/07!	12077.1	12078.1	12079.1	12081.1	

TAB PASS DATA		METH	METHOD I	METHOD II	11 00	METHO	METHOD III
ς. Σ.	TIME	RA-RS	CA-CS	28-48	CA-CS	RA-RS	CA-CS
11090	1400;	18.01	-10.06	36.85	-4.773	18.81	4.7.70
11080	1,500.	-47.79	2,475	-52.35	6.992		7 927
11080	1413,	85.70	23,05	26.50	26.38	38.04	27.09
11080	1706.	19.04	42.48	80.04-	45.89	-15.67	46.33
11113	407.0	-23,56	27,74	-35.77	36.59	-39.10	38.68
11110	100.0	-24.59	38,34	-32.37	47.62	-36.90	49.86
11110	400.0	-33,78	38.13	-31.86	33,87	-24.94	35,87
67111	5.34.0	54.10	71.67	35.28	70.88	23.44	74.24
02111	430.0	-36.64	1001	+5.85-	101.3	-71.52	9.40
11160	140.0	-61.85	70.46	16.49-	96.49	-54.32	69.76
111143	6.00.0	11.75	32,97	4.022	24,58	6.276	26.26
11140	6.004	42.61	51.53	39.48	40.77	36.54	42.32
11140	20004	58.22	33,78	47.41	28,79	57.93	34.04
11140	0.004	19.16	38.90	4.066	36.16	23.01	38.00
11150	900.0	14.57	15.79	-9.119	19,05	17.83	20.14
05111	1700.	-62.99	12.71	-87.91	18,27	-57.68	21.35
	.001	18.45	10.87	31.53	22,37	31.74	22.25
11150	00021	504.6	97.00	6655.	31.98	5.976	30.79
11153	715.0	10.28	41.42	04.18	+6+07-	74.11	*
					3,167	18110°	30.11
11150	415.7	34.32	78.03	22.26	62,06	18.20	67.66
11160	915.0	29.22	16.90	2.539	7,026	13,35	12.80
07111	1015.	-1.35/	15.72	-28.08	5.422	-20.24	5.658
11140	11.28.	21,08	30.412	-17.19	7.077	-2.403	16.50
11130		64.42	19.61	970.0	04.71	21.16	26.92
11180	230.0	42 62	1001	70.05	16.75	19.64	15.94
11140	430.0	-7.781	42,20	19.61	40.53	24.52	44.77
11190	430.0	-166.0	35.05	-142.7	49.79	7.07	47.01
11190	430.0	27.67	18.10	-11.08	9.726	-5.727	14.07
11190	0.010	33.48	10.01	-1			
11230	1102.	23.03	14.01	77.61-	-34.63	10.08	.29.51
11200	1400.	41.06	28.74	27.22	29.10	28.22	417.8
11200	1400.	-28.03	-8.050	10.01		77.07	11.07
11200	1400.	-48.31	47.73	64.01-	49.13	-9.972	45.34
11230	731.0	-77.30	10,18	++.06-	22,51	84.48-	22,32
11230	430.0	-46.82	-42.59	06.19-	-30.54	-57.23	.31.21
1123.	0.000	32.33	-62.58	-24.01	-53,44	10.53	-53,36
11240		26.31	.101.	-33.48	-90,37	56.19	.88.81
	.130.	180.0	23.08	-133.6	-30,19	-124.5	-32.98
11247	1230.	-267.0	-76.13	-220.4	-69.95	-204.0	-59.82
11240	1130,	-31.24	-18.03	-21.42	2,231	-18.49	-8.144
11240	1432.	-118,5	-22.33	-109.5	-2,287	-84.60	.7220
16711	1132.	-33.81	-14.84	-60.83	-18,17	-68.86	-19.07
19711	1435.	-27.52	-4.508	-55.41	-9.660	-62,85	-10.18
27611	15.55	-25.95	-17.34	-37.55	-15.69	-8.697	-17.10
11270	1030	-2.221	-61.25	-13.77	-59.84	36.73	-60.07
11270	1.36.	09.21	65.67	49.584	-30.72	+01.9-	-30.18
11270	1230.	48.17	-12.95	72.61	05.51	27.28	97.
					10171-	06.72	77./1-

METHOD III	CA-CS		6.129	42.68	63 23	17000	90.89	32.56	6.214	53.12	25.53	32 73	-12.44		6.741	3.113	-2.000	-8-926	2076	6380	0070	003.	7.5	19.67	20.10	400		1771.6-	941.5-	21.88	-10.77	-12.29	-25.91	-27.30	
ME	RA-RS		47.31	-22.55	-28.83		38.01	3.261	-62.21	-10.34	.11.16	.9.081	-9.380		.10.6-	22.51	56.95	-233.6	-203.8	-105.6	-105.A	4 - 1		-20 27		-22.36	8.450	2000	-27.30	115.2	31.45	-31.15	+1.1+-	-40.60	-2.711
D II	CA-CS		13.03	36.42	47.29	.0 17	51.00	64.47	400.7	52,95	22,33	32,72	-10.39		99/09	.5710	-3,778	-13,84	-5.975	-11.78	-7.724	151.8	11.81	33.96		-5.688	-8.693			56.33	. 47.4-	-11.73	-24.19	-30.38	-34.75
METHO	RAIRS		75.4.	-30.70	-37.37	29.39	1		14.564	11.60	-15.80	-9.395	-13.29	-13.63	18.71	14.57	24.70	-234.0	-206.6	-101-1	-114.8	-131.4	-166.4	-32.76		-27.53	-9.720	-24.48	104.3	3.001	11.87	74.61-	-43.86	-65.16	-12.25
0 1	CA+CS	12.63	30 4 6	0 0	50.03	75.31	39.81	-4.475		15.01	23122	33.34	-8.913	10.04	1.463	7001	44107	77.91-	-8.495	-77.75	-17.20	9.767	13,38	33,93		-8.166	-3,004	-5.929	21.13	-10.05			00.10	18.74-	-47.89
METHO	RA-RS	72.82	2.510		711.0	63.95	17.86	-17.21	34 87	30.00	*****	18.10	1.63.1	7.229	10.44	20 14		1.701	5.851	11.11-	00.48	-136.3	-172.4	-34.14		17.42-	-15.23	-32,35	118.8	39.19	64.14	. 00	1000	18.75	20.27
	TIME	1130.	432.0	733 0		6.054	930.0	1034.	1130.	1333	1330			800°0	930.0	1000	1230.	1330				1430.	1730.	430.1		0.000	130.0	630.0	1132.	1130.	1230.	1334.	1430		
	C.	11270	12020	12020	13030	63021	12020	12030	12030	12030	12039	12030		12030	12030	12030	12050	12050	12050	12350	12060	13040	2007		12070	1207	13073	2007	12040	12041	12080	12040	12040	12330	

OMNI AB PASS DATA

SERIES	.0000	.0000	0000	0000	0000	.0000	7.00000	.0000	.0000	.000	0	.000	3.000	4.000		00000	7.000	8.000	9.000	000.0	1.000	2.000	3.000	4.000	5.000	000.9	000.	0000	30.000	000	2.000	3.000	4.000	9.000	000.9	37.0000	8.000	.000	000.0	1.000	0	3.000	4.000	5.000	9 300	7.000	9.000	0000.6	000.0
CROSS	98.50	12.40	34.30	53.60	81.40	99.20	262.700	09.49	0+ + + 6	95.20	62.60	85.30	79.10	72.30	249.100	42.10	93.30	04.99	19.80	30.70	31.10	79.30	88.70	95.90	04.50	40.40	07.0	01.00	338.500	83.10	56.90	77.40	45.10	01.70	78.20	317.500	01.00		4 • 30	8.30		5.20	9.10	5.50	06.0	2 . 20	2 . 30	9.70	00 • 0
RANSE	1	7	Ŧ	13959.3	*	7	13942.7	14040.7	1	4	. 666	045.	4166.	3919.		3343.	3942.	4043.	3945.	3911.	4269.	3747.	3974.	3913.	3929.	13972.2		2000	4011.	13908.2	931.	4002.	3957.	3936.	4013.	17	3742		3/32.	948.	4042	3937.	3673.	4048.	3940.	.5+6	4117.	. 600	4018
TIME	1400,00	1500.00	1613.00	734.	607.000	700.000	800.000	534,000	430,000	740.047	500.000	630.000	00	00	0	5	100	200	-	ď	815.010	916		12	000				.00	930.000	0	0.00	500.	5	31.00	.00			•	1230.00	334.	432.0	337.0	434.0	525.n	630.0	030.0	0.	1230.00
40	11081.2	11082.2	111193.2	11084.2	1111111	11112.1	11113.1	11121.2	11122.2	11123.2	-	1 4	+	*	11145.1	*	+	*	75.		11154.1	11155.1	11156.1	11157.1	11158.1	11182.2	11184.2		11193.1	11194.1	1201.	1202.	1 203.	1204.		11232.1	234	316		1236	11237.1	1 2 38	1261	1562	1263	264	1/7	272	5/3

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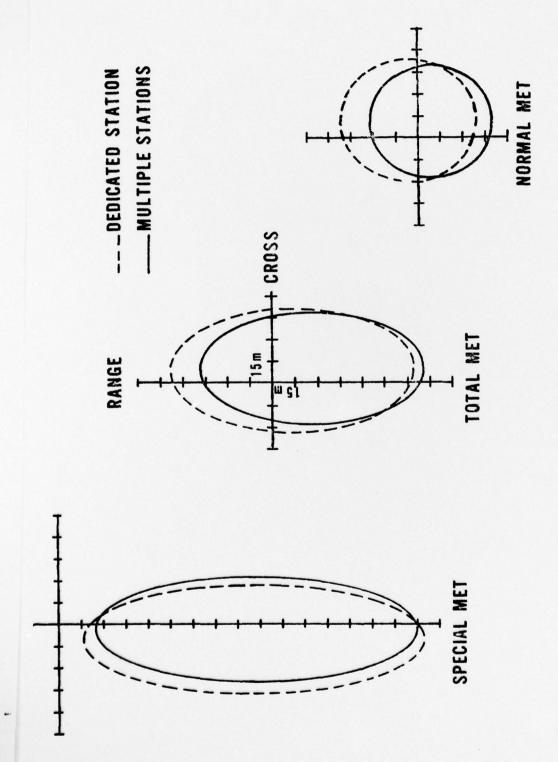


Figure 3. One probable error difference eclipse (actual - simulated).